

**DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION**

S00007LA

WAC

Aeros 40B

June 21, 2000

TYPE CERTIFICATE DATA SHEET NO. S00007LA

This data sheet which is part of Type Certificate No. S00007LA lists the conditions and limitations under which the product for which the type certificate was issued meets the airworthiness requirements of the Federal Aviation Regulations.

Type Certificate Holder Worldwide Aeros Corp.
8411 Canoga Ave.
Canoga Park, CA 91304
U.S.A.

I. Model Aeros 40B Airship. Approved June 21, 2000

Envelope: Volume 88,570 cu. ft.
Maximum Pressure 3.0 in. H₂O
Minimum Pressure 0.8 in. H₂O

Lifting Gas: Helium gas only

Engine Two (2) Continental IO-240-B engines
Type Certificate No. E7SO.

Fuel 100LL minimum grade aviation gasoline.

Engine Limits 2800 rpm for all operations (125 hp)

Propeller and Two (2) MT Propeller Company, MTV-7-D/LD170-12, electric, two position (forward
Propeller Limits and reverse), 3 blade, laminated wood composite structure propellers
Diameter: not over 67.0 in., not under 65.0 in.
No further reduction permitted

Pitch settings at 30 in. station:

Forward: $8.7^{\circ} \pm 2^{\circ}$

Reverse: $-15.3^{\circ} \pm 2^{\circ}$

Static rpm at permissible throttle setting:

Forward: not over 2400, not under 2300

Reverse: not over 2400, not under 2300

No additional tolerance permitted

See NOTE 5

Airspeed Limits

Maximum level Flight Speed

51 m.p.h. IAS

Page No.	1	2	3
Rev. No.	-	-	-

	Maximum Operating Limit Speed	51 m.p.h. IAS
	Design Speed for Maximum Gust Intensity	49 m.p.h. IAS
Center of Gravity (C.G.) Range	See FAA Approved Airship Flight Manual, Report No. 40B-SH	
Datum	<p>The envelope datum is taken at the envelope nose, thus the longitudinal and vertical arms are referred to in envelope station (Env. STA).</p> <p>The car datum (Car STA) is taken at the forward face of the inverted lateral "T" on the floor (below the instrument panel), which is located 7 inches aft of the car nose at floor level, exclusive of the hand rail.</p>	
Maximum Weight	See FAA Approved Airship Flight Manual, Report No. 40B-SH	
Minimum Crew	1 pilot	
No. of Seats	5, (2 at Env. STA 622 (Car STA 26)), 3 at Env. STA 660 (Car STA 64))	
Fuel Capacity	<p>75.9 US gallons total at Env. STA 681 (Car STA 85)</p> <p>70.5 US gallons usable at Env. STA 681 (Car STA 85)</p> <p>See Note 1 for data on weight and balance</p>	
Oil Capacity	<p>6 US quarts at Env. STA 658 (Car STA 62)</p> <p>(3 US quarts usable at Env. STA. 658 (Car STA 62))</p> <p>Each engine</p> <p>See Note 1 for data on weight and balance</p>	
Maximum Operating Altitude	7,500 ft.	
Serial Numbers Eligible	A40B-17 and subsequent.	
Certification Basis	<p>14 CFR part 21, § 21.17(b) , Amendment 21-70, effective September 9,1992</p> <p>Compliance with 14 CFR part 21, § 21.17(b) has been shown utilizing the provisions of Advisory Circular 21.17-1A, change 1, dated October 30, 1992. The airworthiness requirements met under this provision are the Airship Design Criteria (ADC), FAA P-8110-2, dated November 2, 1987, as amended by Change 1, dated July 24, 1992 and Change 2, dated February 6, 1995. See Note 6</p> <p>Application for Type Certificate dated October 17, 1997</p> <p>Type Certificate No. S00007LA issued June 21, 2000</p> <p>Noise Control Act Finding dated March 2, 2000</p> <p>Finding of No Significant Impact (FONSI) dated March 2, 2000</p>	
Production Basis	None. Prior to original certification of each aircraft, a FAA representative must perform a detailed inspection for workmanship, materials, and conformity with the approved technical data, and a check of the flight characteristics.	
Equipment	The basic required equipment as prescribed in the applicable airworthiness regulations (see Certification Basis) must be installed in the airship for certification.	

In addition to the above required equipment, the following equipment is also required: FAA-Approved Airship Flight Manual, Report 40B-SH, dated June 15, 2000, or later FAA approved revision.

NOTES:

Note 1

A current weight and balance report, including a list of equipment included in the certificated empty weight and loading instructions when necessary, must be provided for each airship at the time of original certification. The certificated empty weight and the corresponding center of gravity location must include full oil (11.3 lbs. at Env. STA 658 (Car STA 62)) and unusable fuel (32.4 lbs. at Env. STA. 681 (Car STA 85)).

- Note 2 The placards specified in the Approved Airship Flight Manual must be displayed at the appropriate locations.
- Note 3 In accordance with the Airship Design Criteria (ADC), FAA P-8110-2, Paragraph 7.12; Instructions for Continued Airworthiness acceptable to the Administrator must be completed prior to delivery of the first airship or issuance of a standard certificate of airworthiness, whichever occurs later.
- Note 4 Reserved
- Note 5 MT propeller Data Sheet No. P-20BO provides information on the propeller. The electric pitch control components are described in Worldwide Aeros Corp. drawing numbers A40B-06-28-00 and A40B-06-28-01. See WAC Specification WAC 40B-010 for propeller pitch adjustment procedure.
- Note 6 ADC Para. 3.24, change 2, specifies a longitudinal load on the landing gear of $.25n W_l$. A longitudinal load of $.25n W_{SH}$ was used in the design of the Aeros 40B landing gear and supporting structure.
- Note 7 Propellers must be marked in accordance with WAC Specification WAC 40B-011.

.....END.....